

Optimization of composite structures

A topology and shape sensitivity analysis

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Main topics

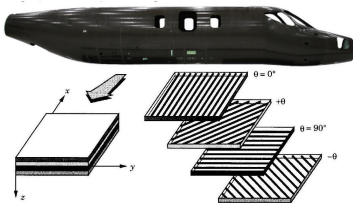
1) Composite optimal design via a level-set method

2) Anisotropic topological derivative

Main topics

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(a) Multi-layered composites

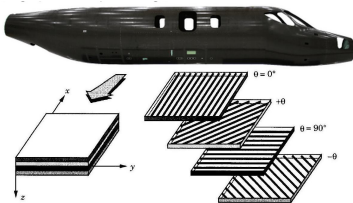


2) Anisotropic topological derivative

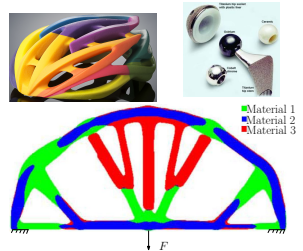
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1) Composite optimal design via a level-set method

(a) Multi-layered composites



(b) Multi-phase materials

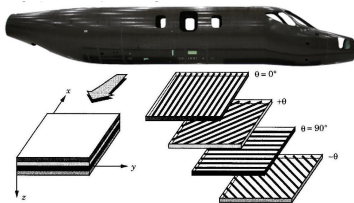


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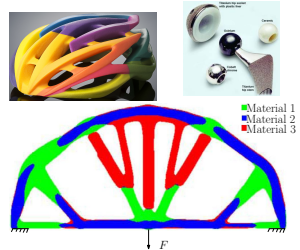
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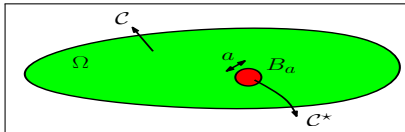
(a) Multi-layered composites



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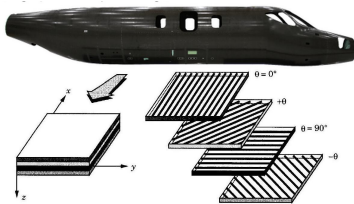


$$DJ = \lim_{a \rightarrow 0} \frac{J(a) - J(0)}{h(a)}$$

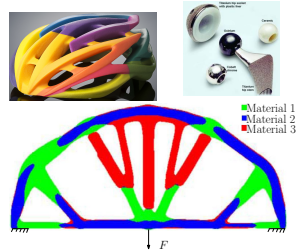
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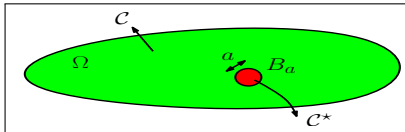
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2) Anisotropic topological derivative



$$DJ = \lim_{a \rightarrow 0} \frac{J(a) - J(0)}{h(a)}$$

- Topology optimization.
- Inverse problems.

Outline

- 1 Overview on topology optimization & the level-set method
- 2 Optimal design of composite materials
- 3 Topological derivative
- 4 Conclusions and perspectives

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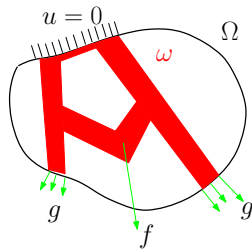
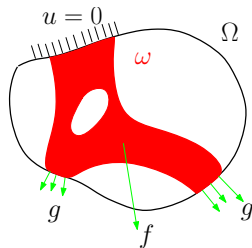
Topology optimization

A topology optimization problem can be cast as follows:

Find the stiffest structure $\omega \subset \Omega$
 belonging to an admissible set \mathcal{U}_{ad} ,
 which solves the problem

$$\min_{\omega \subset \Omega, \omega \in \mathcal{U}_{ad}} J(\omega, u(\omega)),$$

where $u(\omega)$ is the displacement field arising in ω due to prescribed excitations (f, g) .



Topology optimization methods

Density based	Boundary variations
Homogenization [Murat and Tartar, 1985], [Bendsøe and Kikuchi, 1988]	Level set+Shape sensitivity analysis [Allaire et al., 2002], [Wang et al., 2003]
SIMP [Bendsøe, 1995]	Phase field [Chambolle and Bourdin, 2000]

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Comparative advantages

- Clear contour of the shape.
- Easy definition of geometrical parameters.

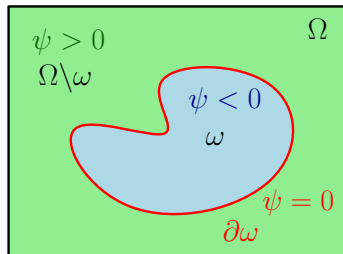
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- Definition of the level set function $\psi : \Omega \rightarrow \mathbb{R}$, with $\Omega \subset \mathbb{R}^d$ ($d = 2, 3$)

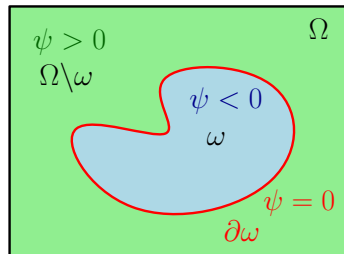
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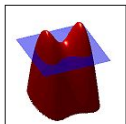
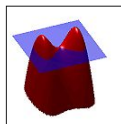
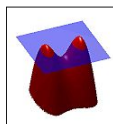
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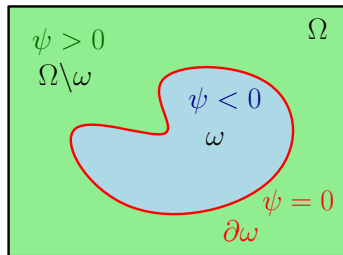
- Evolution of the shape $\omega(t), t \in \mathbb{R}$
Level-set function $\psi(x, t)$

 t_0  t_1  t_2

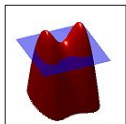
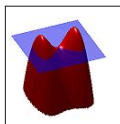
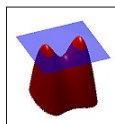
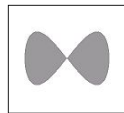
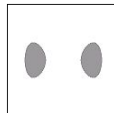
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 t_0  t_1  t_2  t_0  t_1  t_2 Shape $\omega(t)$

The Level-set method

- If the shape evolves according to a velocity field $\theta(x, t)$ defined on Ω with normal component $\mathcal{V}(x, t)$

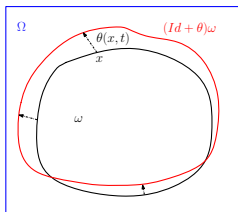


Figure: Deformation of ω

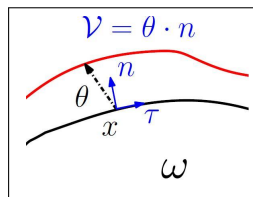


Figure: Zoom around x

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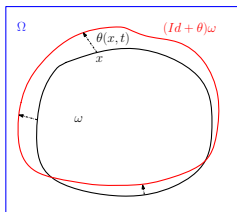


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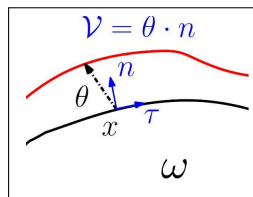


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- One can show that the evolution equation satisfied by ψ reads

$$\left\{ \begin{array}{ll} \frac{\partial \psi}{\partial t} + \mathcal{V} |\nabla \psi| = 0 & x \in \Omega, t > 0, \\ \psi = \psi_0 & x \in \Omega, t = 0, \\ \frac{\partial \psi}{\partial n} = 0 & x \in \partial \Omega, t > 0. \end{array} \right.$$

Shape sensitivity analysis

Definition [Hadamard, 1908], [Simon and Murat, 1976]

Consider the following type of variations of $\omega \subset \Omega$

$$(Id + \theta)(\omega) := \{x + \theta(x) \text{ for } x \in \omega\}, \theta \in W^{1,\infty}(\Omega; \mathbb{R}^d)$$

The **shape derivative** of a function $J(\omega)$ is defined as the Fréchet derivative in $W^{1,\infty}(\Omega; \mathbb{R}^d)$ at 0 of the application $\theta \rightarrow J((Id + \theta)\omega)$, i.e.

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Coupling the level set method and shape derivative

- The objective is to find a normal descent direction \mathcal{V}^* such that

$$\begin{cases} \psi(t_0, \cdot) & \equiv \omega \\ \psi(t_0 + \Delta t, \cdot) & \equiv \omega^{\Delta t} \end{cases} \Rightarrow J(\omega^{\Delta t}) - J(\omega) < 0$$

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Indeed, for Δt small enough

$$\begin{aligned} J(\omega^{\Delta t}) - J(\omega) &= \Delta t \times J'(\omega)(\theta^*) + o(\Delta t), \quad \theta^* = \mathcal{V}^* n \\ &= \Delta t \times l(\mathcal{V}^*) + o(\Delta t) \\ &= -\Delta t \times \|\mathcal{V}^*\|_{H^1(\Omega)}^2 + o(\Delta t) < 0 \end{aligned}$$

The level set method for topology optimization

- Optimization of a triangular car suspension (courtesy of G. Allaire).

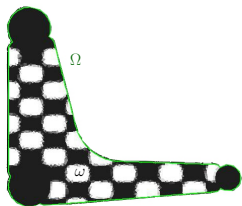


Figure: Iteration 1

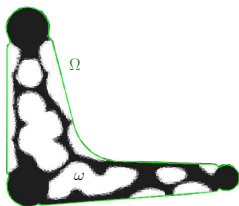


Figure: Iteration 26

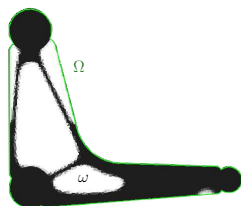


Figure: Iteration 100

Linear elasticity:

$$\begin{cases} -\operatorname{div}(Ce(u)) = 0 & \text{in } \Omega \\ u = 0 & \text{on } \Gamma_D \\ Ce(u) \cdot n = g & \text{on } \Gamma_N \end{cases}$$

$$\min_{\omega \subset \Omega, |\omega|=c} J(\omega), \quad J(\omega) = \int_{\Gamma_N} g \cdot u ds$$

$$J'(\omega)(\theta) = - \int_{\partial\omega} \theta \cdot n (Ce(u) : e(u)) dx$$

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Multi-phase optimal design

- Design variable: shape of each phase (n level-sets $\rightarrow 2^n$ materials)

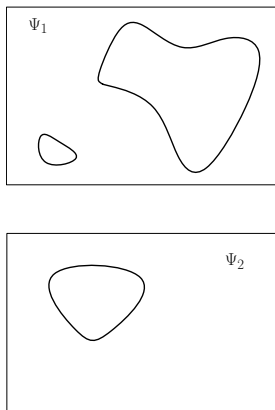


Figure: Level-set functions Ψ_1, Ψ_2

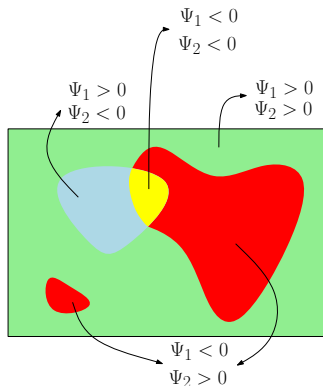


Figure: Multi-phase material via superposition of Ψ_1, Ψ_2

Multi-layered optimal design

- Design variables: shape of each ply Ψ_i (continuous) & stacking sequence ξ (discrete)

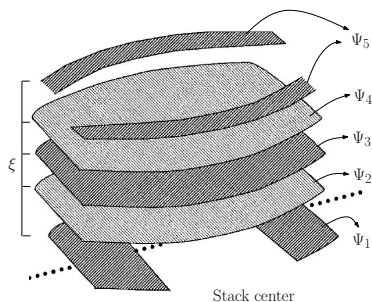


Figure: Multi-layered configuration

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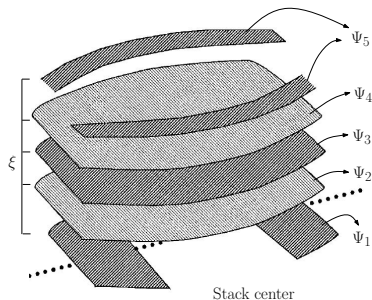


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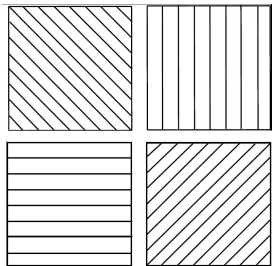


Figure: Orthotropic plies.
Reinforced fibers in orientations $-45^\circ, 90^\circ, 0^\circ, 45^\circ$

The buckling problem

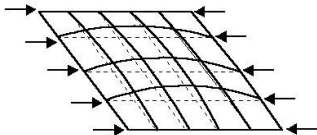


Figure: Buckling under compression



Figure: Buckled wing

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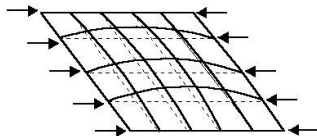
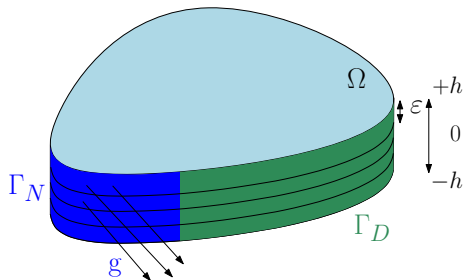


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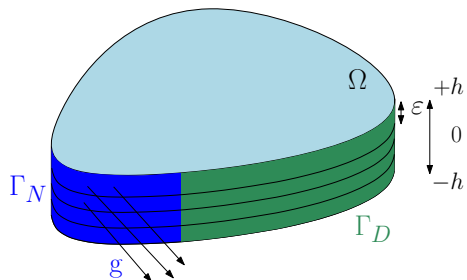


- Ω : Hold-all domain
- g : in-plane loads
- Γ_N : Neumann BC
- Γ_D : Dirichlet BC
- ε : Thickness of each ply
- $2h$: Total thickness laminate
- A : extensional stiffness
- D : bending stiffness

The buckling problem

$$\mathcal{A} = 2\varepsilon \sum_{i=1}^N \left(\chi^i \mathcal{A}^i + (1 - \chi^i) \mathcal{A}^0 \right), \quad \chi^i : \text{characteristic function}$$

$$\mathcal{D} = \frac{2\varepsilon^3}{3} \sum_{i=1}^N \left(i^3 - (i-1)^3 \right) \left(\chi^i \mathcal{A}^i + (1 - \chi^i) \mathcal{A}^0 \right)$$



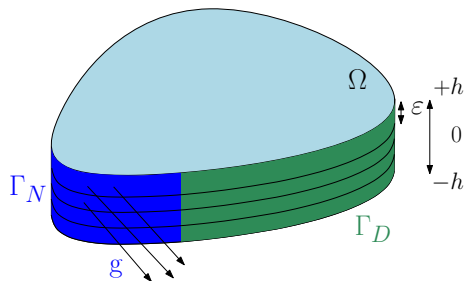
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The multi-layered design problem becomes a multi-phase design problem.



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The linearized buckling problem

Definition (von Kármán plate model)

Let $u \in H^1(\Omega; \mathbb{R}^2)$ be the **in-plane displacement** and solves

$$\begin{cases} -\operatorname{div}(\mathcal{A}e(u)) = 0 & \text{in } \Omega, \\ u = 0 & \text{on } \Gamma_D, \\ \mathcal{A}e(u) \cdot n = 2hg, & \text{on } \Gamma_N. \end{cases}$$

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We chose $\lambda = \lambda_1$, the smallest positive eigenvalue, as the critical buckling load.

The linearized buckling problem

Definition (von Kármán plate model)

Let $w \in H^2(\Omega)$ be the **vertical displacement**. Consider the eigenvalue problem

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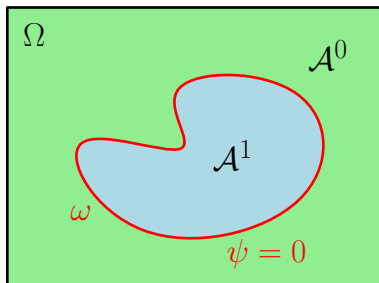
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Sharp interface formulation

$$\mathcal{A} = \mathcal{A}^1 + H(\psi)(\mathcal{A}^0 - \mathcal{A}^1)$$

$$\begin{cases} \mathcal{A}^1, \mathcal{A}^0 : \text{Elastic phases} \\ H : \text{Heaviside function.} \end{cases}$$



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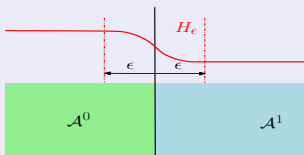
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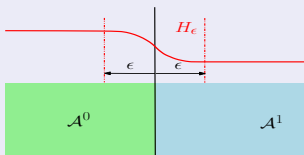
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Signed distance function:

$$d_\omega(x) = \begin{cases} -d(x, \partial\omega) & \text{if } x \in \omega, \\ 0 & \text{if } x \in \partial\omega, \\ d(x, \partial\omega) & \text{if } x \in \Omega \setminus \bar{\omega}, \end{cases}$$

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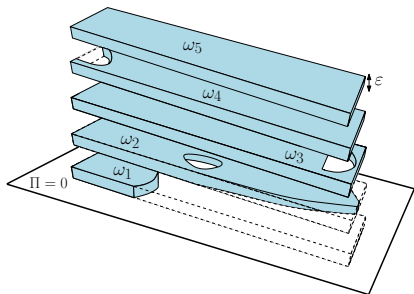


Figure: Each ply has its own shape ω_i

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Figure: Transversal cut. Multi-phase structure.

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Proposition

The mixed lightweight design composite problem can be equivalently written as

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Proposition [Allaire, Dapogny, Delgado & Michailidis (2014)]

I. Continuous sharp interface

The shape derivative of J reads

$$J'(\omega)(\theta) = - \int_{\partial\omega} D(u, p) \theta \cdot n ds, \quad \text{with}$$

$$D(u, p) = -\sigma(p)_{nn} : \llbracket e(u)_{nn} \rrbracket - 2\sigma(p)_{n\tau} : \llbracket e(u)_{n\tau} \rrbracket + \llbracket \sigma(u)_{\tau\tau} \rrbracket : e(p)_{\tau\tau}.$$

where $\sigma(v) = \mathcal{A}e(v)$ and p is the adjoint state, solution of

$$-\operatorname{div}(\mathcal{A}e(p)) = -j'(x, u) \text{ in } \Omega, \quad \mathcal{A}e(p) \cdot n = 0 \text{ on } \Gamma_N, \quad p = 0 \text{ on } \Gamma_D.$$

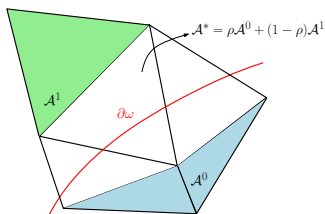
Multi-phase shape sensitivity analysis

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II. Discrete sharp interface

Let Ω_h be a conformal simplicial mesh of Ω and u_h, p_h the finite element approximations of u, p , respectively. Suppose that $\partial\omega$ is never aligned with the face of an element of Ω_h . Then the discrete shape derivative of $J_h(\omega) = \int_{\Omega_h} j(x, u_h) dx$ reads

$$J'_h(\omega)(\theta) = - \int_{\partial\omega} [[\mathcal{A}]] e(u_h) : e(p_h) \theta \cdot n \, ds.$$



$$\sigma^0(u_h^0) \cdot n^0 \neq \sigma^1(u_h^1) \cdot n^1$$

$$u_h^0 = u_h^1$$

$$\mathcal{A}^* = \rho \mathcal{A}^0 + (1 - \rho) \mathcal{A}^1$$

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III. Smooth interface

Consider the smooth interface frame $\mathcal{A} = \mathcal{A}^1 + H_\epsilon(d_\omega)(\mathcal{A}^0 - \mathcal{A}^1)$, with d_ω the signed distance function. Then the function J is shape differentiable in the sense of Gâteaux, and

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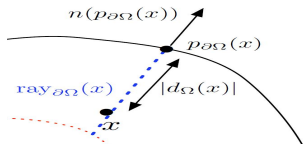
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Proposition (Convergence)

$$\lim_{\epsilon \rightarrow 0} J'_\epsilon(\omega)(\theta) = J'(\omega)(\theta), \quad \forall \theta \in W^{1,\infty}(\Omega; \mathbb{R}^2)$$

Test case formulation: Composite fuselage skin panel

- $\Omega = \{x \in [0, 2] \times [0, 1]\}$. Symmetric multi-layered plate (16 plies).
- Each ply is composed of two phases (one of them “void”).
- A shear load g is applied.

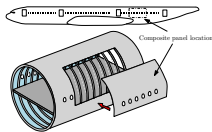


Figure: Approximative flat model.



Figure: Composite test case.

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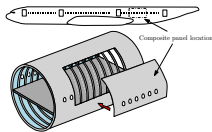


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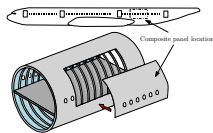


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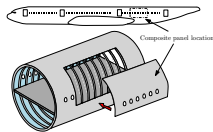


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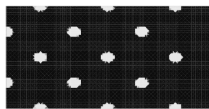
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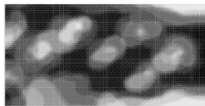
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- Continuous algorithm: Sequential linear programming.
- Discrete algorithm: Outer approximation method.

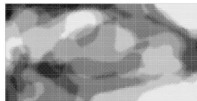
Test case results



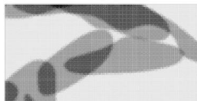
t0



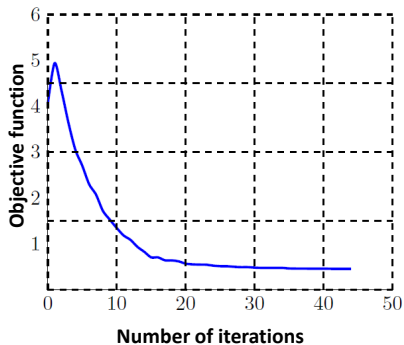
t1



t2



t3



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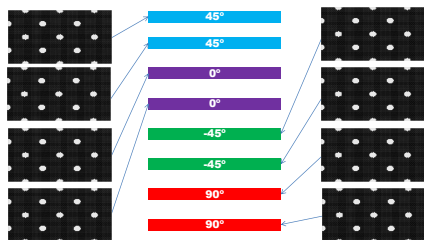


Figure: Initial configuration

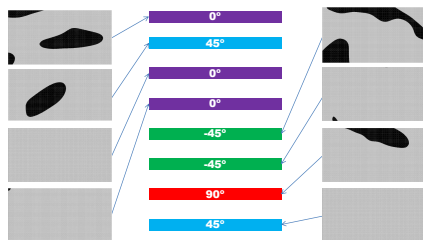


Figure: Final configuration (45 iter.)

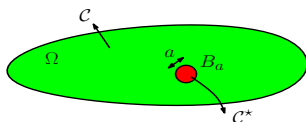
Outline

- 1 Overview on topology optimization & the level-set method
- 2 Optimal design of composite materials
- 3 Topological derivative**
- 4 Conclusions and perspectives

Brief overview of the state of the art

Theory	*[Sokolowski & Zochowski99] *[Masmoudi98]
--------	--

- Purpose: Measures the sensibility of a cost function J defined on $\Omega \subset \mathbb{R}^d$ w.r.t. the creation of a virtual inclusion B_a of size $a \ll 1$



$$\begin{cases} C : \text{Physical properties } \Omega \\ C^* : \text{Physical properties } B_a \end{cases}$$

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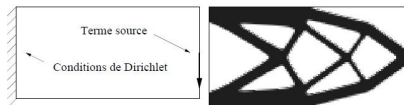


Figure: Topology Optimization

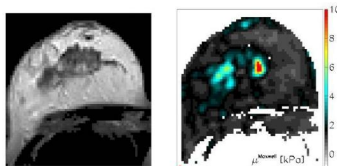


Figure: Inverse problems [Guz06]

Brief overview of the state of the art

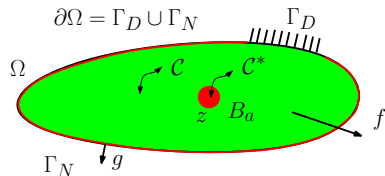
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Novelty of this work	<ul style="list-style-type: none"> *General anisotropic framework *Stress-based functionals

The topological derivative in 3D elastostatics

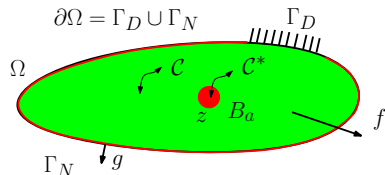
- Let u_a, u be the perturbed and non-perturbed elastic displacements within Ω .



$$\left\{ \begin{array}{l} \mathcal{C} : \text{Elastic tensor in } \Omega \\ \mathcal{C}^* : \text{Elastic tensor in } B_a \\ f, g : \text{Volume \& Surface loads} \\ z : \text{Center of } B_a \end{array} \right.$$

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- where $W = \{v \in H^1(\Omega; \mathbb{R}^3) | v = 0 \text{ on } \Gamma_D \subset \partial\Omega\}$ and $u_a, u \in W$ are solutions of:

$$\int_{\Omega} \nabla u : C_a : \nabla v dx = \int_{\Omega} f \cdot v dx + \int_{\Gamma_N} g \cdot v ds, \quad \forall v \in W,$$

for $C_a = C^* \chi_{B_a} + C(1 - \chi_{B_a})$ and $C_0 = C$, respectively.

The topological derivative in 3D elastostatics

- Let u_a, u be the perturbed and non-perturbed elastic displacements within Ω .
- The cost function will take the form

$$J(u_a, \nabla u_a) := \int_{\Omega \setminus B_a} \phi(x, u_a, \nabla u_a) dx + \int_{B_a} \phi^*(x, u_a, \nabla u_a) dx$$

where ϕ, ϕ^* are twice differentiable and $\partial_{ij}^2 \phi, \partial_{ij}^2 \phi^* \in C^{0,\alpha}$, $\alpha \in (0, 1)$.

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Definition

Let the cost functional J be and assume that it can be expanded in the form

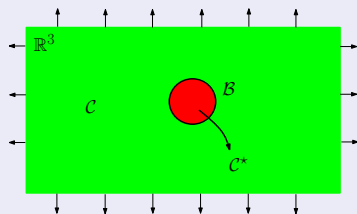
$$J(u_a, \nabla u_a) - J(u, \nabla u) = a^3 DJ(z) + o(a^3).$$

Then $DJ(z)$ is called the **topological derivative** of J at $z \in \Omega$.

How to compute u_a for a small inhomogeneity?

Lemma

Consider the auxiliary problem of an inhomogeneity $(\mathcal{B}, \mathcal{C}^*)$ centered at the origin, embedded in an infinite elastic medium $(\mathbb{R}^3, \mathcal{C})$ subjected to a uniform remote strain $\nabla u(z), z \in \Omega$.



$$\begin{cases} u_{\mathcal{B}} : \text{perturbed displacement in } \mathbb{R}^3 \\ u^{\infty} : \text{unperturbed displacement in } \mathbb{R}^3 \\ u^{\infty} = \nabla u(z) \cdot x, x \in \mathbb{R}^3 \end{cases}$$

Suppose that $B_a = z + a\mathcal{B}$, then the following near-field development stands

$$u_a = u + a(u_{\mathcal{B}} - u^{\infty}) \left(\frac{x - z}{a} \right) + o(a).$$

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Definition

Let $u_{\mathcal{B}}$ denote the perturbed solution of the free-space transmission problem. **The elastic moment tensor (EMT) \mathcal{E}** is defined for any value of $\nabla u(z)$ as

$$\mathcal{E} : \nabla u(z) = \int_{\mathcal{B}} (\mathcal{C}^* - \mathcal{C}) : \nabla u_{\mathcal{B}} dx.$$

Main result

Proposition [Delgado and Bonnet, 2014 (review)]

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 & + \int_{\mathbb{R}^3 \setminus \mathcal{B}} \mathcal{G}(z, \nabla v_{\mathcal{B}}(x)) dx + \int_{\mathcal{B}} \mathcal{G}^*(z, \nabla v_{\mathcal{B}}(x)) dx.
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- Similar results: [Amstutz et al., 2012], [Schneider and Andrä, 2013].

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- New terms appearing in DJ are due to the dependence of J on ∇u_a .
- In particular the evaluation of the term

$$\int_{\mathbb{R}^3 \setminus \mathcal{B}} \mathcal{G}(z, \nabla v_{\mathcal{B}}(x)) dx$$

is **numerically expensive** to integrate.

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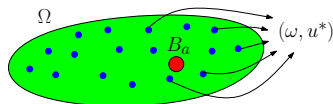


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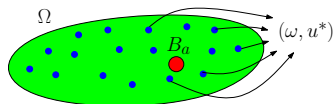
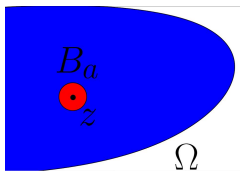
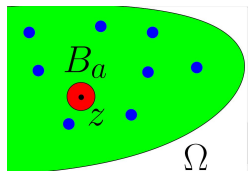


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- Some examples: Magnetic resonance elastography [Yuan et al., 2012], Digital image correlation via X-ray tomography [Bay et al., 1999].

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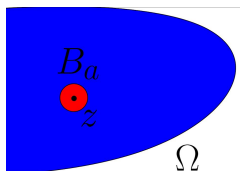


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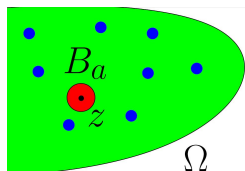


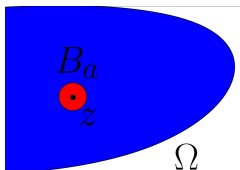
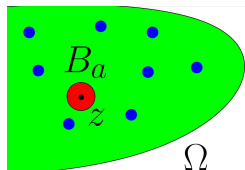
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- The goal is to find the places z where DJ attains the **most negative values**.

First application: 2D framework, $\omega = \Omega$

- Objective: Identify each inclusion.

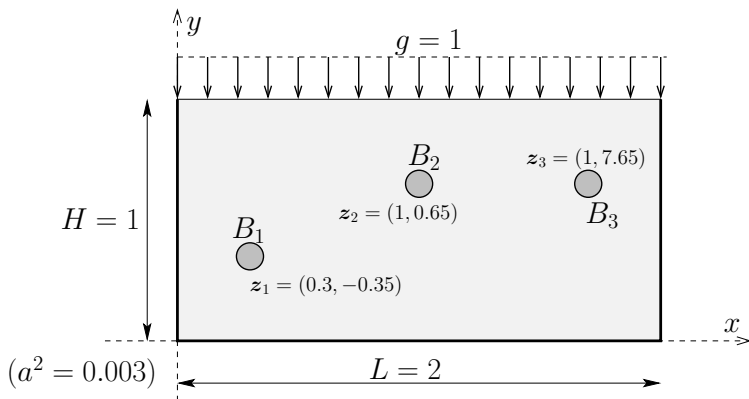


Figure: Orthotropic medium \mathcal{C} ($E_1 = 0.1, E_2 = 1, \nu_{12} = 0.3, G_{12} = 0.03$) with three inclusions: B_1, B_3 are softer ($\mathcal{C}^* = 0.5\mathcal{C}$) while B_2 is harder ($\mathcal{C}^* = 2\mathcal{C}$).

First application: Results

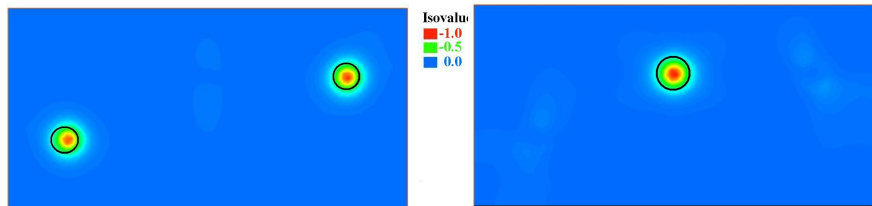


Figure: Values of DJ using as test material $C^* = 0.5C$.

Figure: Values of DJ using as test material $C^* = 2C$.

Second application: 3D composite framework, $\omega \subset \Omega$

- Objective: Detect the failure point

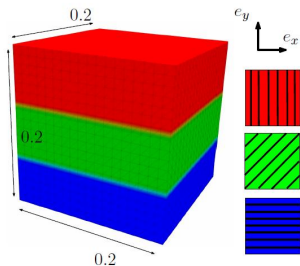


Figure: Multi-layered cube made of three orthotropic materials $\mathcal{C}_1, \mathcal{C}_2, \mathcal{C}_3$. Failure point $\mathcal{C}^* = 10^{-5} \times \mathcal{C}$

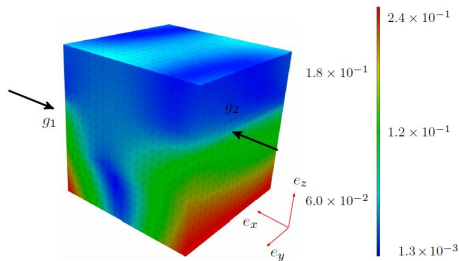


Figure: Elastic displacements after the application of the loads g_1, g_2 .

Second application: Results

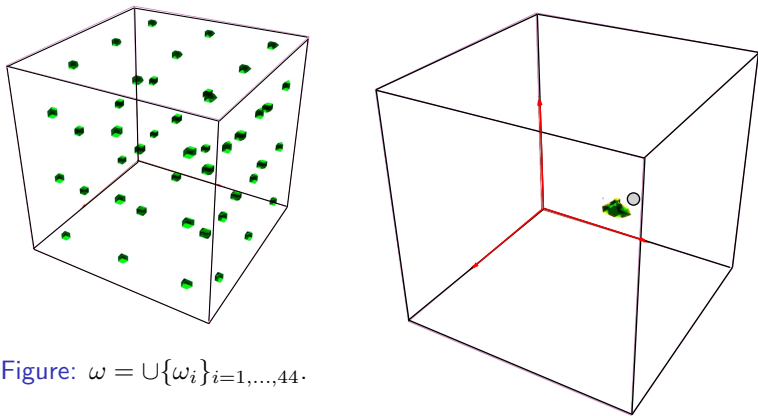


Figure: $\omega = \cup\{\omega_i\}_{i=1,\dots,44}$.

Green: DJ iso-surfaces around its minimum value -1.3×10^{-3} .

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Second application: Results

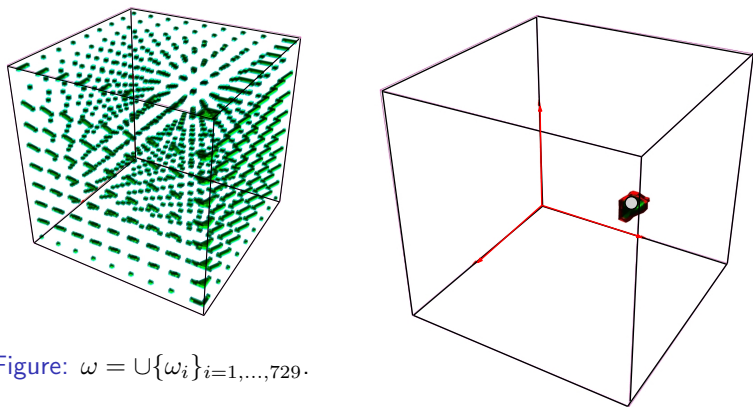


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- The level-set method was extended to a multi-phase & multi-layered composite framework.
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- The topological derivative for anisotropic materials was studied for a large class of criteria.

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- Perform a robustness and sensitivity analysis of the topological derivative for non-destructive control.
- Extend the computation of the topological derivative to an elastodynamic framework.

Thank you for your attention!

Communications

- G. ALLAIRE, C. DAPOGNY, G. DELGADO, AND G. MICHAILIDIS. *Mutli-phase structural optimization via a level-set method*. ESAIM: Control, Optimisation and Calculus of Variations (2014).
- M. BONNET, G. DELGADO. *The topological derivative in anisotropic elasticity*. Quarterly Journal of Mechanics and Applied Mathematics (2013).
- G. DELGADO, M. BONNET. *The topological derivative of stress-based cost functionals in anisotropic elasticity*. Submitted (2014).
- G. ALLAIRE, G. DELGADO. *Stacking sequence and shape optimization of laminated composite plates via a level-set method*. In preparation (2014).

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